

# FREQUENCY RESPONSE FUNCTION FOR BEAM BENDING WITH ENFORCED ACCELERATION VIA FINITE ELEMENT ANALYSIS

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## Variables

A	Enforced acceleration amplitude
M	Mass matrix
K	Stiffness matrix
F	Applied forces
F <sub>d</sub>	Forces at driven nodes
F <sub>f</sub>	Forces at free nodes
I	Identity matrix
Π	Transformation matrix
u	Displacement vector
u <sub>d</sub>	Displacements at driven nodes, time domain
u <sub>f</sub>	Displacements at free nodes, time domain
$\hat{U}_d$	Displacements at driven nodes, frequency domain
$\hat{U}_f$	Displacements at free nodes, frequency domain

## Derivation

The following method is adapted from Reference 1.

The equation of motion for a multi-degree-of-freedom system is

$$[M][\ddot{u}] + [K][u] = F \quad (1)$$

The displacement vector is

$$[u] = \begin{bmatrix} u_d \\ u_f \end{bmatrix} \quad (2)$$

Partition the matrices and vectors as follows

$$\begin{bmatrix} M_{dd} & M_{df} \\ M_{fd} & M_{ff} \end{bmatrix} \begin{bmatrix} \ddot{u}_d \\ \ddot{u}_f \end{bmatrix} + \begin{bmatrix} K_{dd} & K_{df} \\ K_{fd} & K_{ff} \end{bmatrix} \begin{bmatrix} u_d \\ u_f \end{bmatrix} = \begin{bmatrix} F_d \\ F_f \end{bmatrix} \quad (3)$$

Create a transformation matrix such that

$$\begin{bmatrix} u_d \\ u_f \end{bmatrix} = \Pi \begin{bmatrix} u_d \\ u_w \end{bmatrix} \quad (4)$$

$$\Pi = \begin{bmatrix} I & 0 \\ T_1 & T_2 \end{bmatrix} \quad (5)$$

$$\begin{bmatrix} M_{dd} & M_{df} \\ M_{fd} & M_{ff} \end{bmatrix} \Pi \begin{bmatrix} \ddot{u}_d \\ \ddot{u}_w \end{bmatrix} + \begin{bmatrix} K_{dd} & K_{df} \\ K_{fd} & K_{ff} \end{bmatrix} \Pi \begin{bmatrix} u_d \\ u_w \end{bmatrix} = \begin{bmatrix} F_d \\ F_f \end{bmatrix} \quad (6)$$

Premultiply by  $\Pi^T$ ,

$$\Pi^T \begin{bmatrix} M_{dd} & M_{df} \\ M_{fd} & M_{ff} \end{bmatrix} \Pi \begin{bmatrix} \ddot{u}_d \\ \ddot{u}_w \end{bmatrix} + \Pi^T \begin{bmatrix} K_{dd} & K_{df} \\ K_{fd} & K_{ff} \end{bmatrix} \Pi \begin{bmatrix} u_d \\ u_w \end{bmatrix} = \Pi^T \begin{bmatrix} F_d \\ F_f \end{bmatrix} \quad (7)$$

Transform the equation of motion to uncouple the stiffness matrix so that the resulting stiffness matrix is

$$\begin{bmatrix} \hat{K}_{dd} & 0 \\ 0 & \hat{K}_{ww} \end{bmatrix} \quad (8)$$

$$\Pi^T K \Pi = \begin{bmatrix} I & T_1^T \\ 0 & T_2^T \end{bmatrix} \begin{bmatrix} K_{dd} & K_{df} \\ K_{fd} & K_{ff} \end{bmatrix} \begin{bmatrix} I & 0 \\ T_1 & T_2 \end{bmatrix} \quad (9)$$

$$\Pi^T K \Pi = \begin{bmatrix} I & T_1^T \\ 0 & T_2^T \end{bmatrix} \begin{bmatrix} K_{dd} + K_{df} T_1 & K_{df} T_2 \\ K_{fd} + K_{ff} T_1 & K_{ff} T_2 \end{bmatrix} \quad (10)$$

$$\Pi^T K \Pi = \begin{bmatrix} K_{dd} + K_{df} T_1 + T_1^T (K_{fd} + K_{ff} T_1) & K_{df} T_2 + T_1^T K_{ff} T_2 \\ T_2^T (K_{fd} + K_{ff} T_1) & T_2^T (K_{ff} T_2) \end{bmatrix} \quad (11)$$

$$\Pi^T K \Pi = \begin{bmatrix} K_{dd} + K_{df} T_1 + T_1^T (K_{fd} + K_{ff} T_1) & (K_{df} + T_1^T K_{ff}) T_2 \\ T_2^T (K_{fd} + K_{ff} T_1) & T_2^T (K_{ff} T_2) \end{bmatrix} \quad (12)$$

$$\Pi^T K \Pi = \begin{bmatrix} K_{dd} + T_1^T K_{fd} + (K_{df} + T_1^T K_{ff}) T_1 & (K_{df} + T_1^T K_{ff}) T_2 \\ T_2^T (K_{fd} + K_{ff} T_1) & T_2^T (K_{ff} T_2) \end{bmatrix} \quad (13)$$

Let

$$T_2 = I \quad (14)$$

$$\Pi^T \mathbf{K} \Pi = \begin{bmatrix} \mathbf{K}_{dd} + \mathbf{T}_1^T \mathbf{K}_{fd} + (\mathbf{K}_{df} + \mathbf{T}_1^T \mathbf{K}_{ff}) \Gamma_1 & (\mathbf{K}_{df} + \mathbf{T}_1^T \mathbf{K}_{ff}) \\ (\mathbf{K}_{fd} + \mathbf{K}_{ff} \mathbf{T}_1) & \mathbf{K}_{ff} \end{bmatrix} = \begin{bmatrix} \hat{\mathbf{K}}_{dd} & 0 \\ 0 & \hat{\mathbf{K}}_{ww} \end{bmatrix} \quad (15)$$

$$\mathbf{K}_{df} + \mathbf{T}_1^T \mathbf{K}_{ff} = 0 \quad (16)$$

$$\mathbf{T}_1^T = -\mathbf{K}_{df} \mathbf{K}_{ff}^{-1} \quad (17)$$

$$\mathbf{T}_1 = -\mathbf{K}_{ff}^{-1} \mathbf{K}_{fd} \quad (18)$$

$$\Pi = \begin{bmatrix} \mathbf{I}_{dd} & 0 \\ \mathbf{T}_1 & \mathbf{I}_{ff} \end{bmatrix} \quad (19)$$

$$\hat{\mathbf{K}}_{dd} = \mathbf{K}_{dd} + \mathbf{T}_1^T \mathbf{K}_{fd} + (\mathbf{K}_{df} + \mathbf{T}_1^T \mathbf{K}_{ff}) \Gamma_1 \quad (20)$$

$$\hat{\mathbf{K}}_{ww} = \mathbf{K}_{ff} \quad (21)$$

$$\Pi^T \mathbf{M} \Pi = \begin{bmatrix} \mathbf{I}_{dd} & \mathbf{T}_1^T \\ 0 & \mathbf{I}_{ff} \end{bmatrix} \begin{bmatrix} \mathbf{M}_{dd} & \mathbf{M}_{df} \\ \mathbf{M}_{fd} & \mathbf{M}_{ff} \end{bmatrix} \begin{bmatrix} \mathbf{I}_{dd} & 0 \\ \mathbf{T}_1 & \mathbf{I}_{ff} \end{bmatrix} \quad (22)$$

By similarity, the transformed mass matrix is

$$\begin{bmatrix} \hat{m}_{dd} & \hat{m}_{dw} \\ \hat{m}_{wd} & \hat{m}_{ww} \end{bmatrix} = \begin{bmatrix} M_{dd} + T_1^T M_{fd} + (M_{df} + T_1^T M_{ff}) T_1 & (M_{df} + T_1^T M_{ff}) \\ (M_{fd} + M_{ff} T_1) & M_{ff} \end{bmatrix} \quad (23)$$

$$\begin{bmatrix} \hat{F}_d \\ \hat{F}_w \end{bmatrix} = \begin{bmatrix} I_{dd} & T_1 \\ 0 & I_{ff} \end{bmatrix} \begin{bmatrix} F_d \\ F_f \end{bmatrix} \quad (24)$$

$$\begin{bmatrix} \hat{F}_d \\ \hat{F}_w \end{bmatrix} = \begin{bmatrix} I_{dd} F_d + T_1 F_f \\ I_{ff} F_f \end{bmatrix} \quad (25)$$

$$\begin{bmatrix} \hat{F}_d \\ \hat{F}_w \end{bmatrix} = \begin{bmatrix} F_d + T_1 F_f \\ F_f \end{bmatrix} \quad (26)$$

$$\begin{bmatrix} \hat{m}_{dd} & \hat{m}_{dw} \\ \hat{m}_{wd} & \hat{m}_{ww} \end{bmatrix} \begin{bmatrix} \ddot{u}_d \\ \ddot{u}_w \end{bmatrix} + \begin{bmatrix} \hat{K}_{dd} & 0 \\ 0 & \hat{K}_{ww} \end{bmatrix} \begin{bmatrix} u_d \\ u_w \end{bmatrix} = \begin{bmatrix} \hat{F}_d \\ \hat{F}_w \end{bmatrix} \quad (27)$$

$$\hat{m}_{wd} \ddot{u}_d + \hat{m}_{ww} \ddot{u}_w + \hat{K}_{ww} u_w = \hat{F}_w \quad (28)$$

The equation of motion is thus

$$\hat{m}_{ww} \ddot{u}_w + \hat{K}_{ww} u_w = \hat{F}_w - \hat{m}_{wd} \ddot{u}_d \quad (29)$$

Assume that the external forces are zero.

$$\hat{m}_{ww} \ddot{u}_w + \hat{K}_{ww} u_w = -\hat{m}_{wd} \ddot{u}_d \quad (30)$$

Consider the homogeneous form of equation (30).

$$\hat{m}_{ww} \ddot{u}_w + \hat{K}_{ww} u_w = \bar{0} \quad (31)$$

Seek a solution of the form

$$\bar{u}_w = \bar{q} \exp(j\omega t) \quad (32)$$

The  $q$  vector is the generalized coordinate vector.

Note that

$$\dot{\bar{u}} = j\omega \bar{q} \exp(j\omega t) \quad (33)$$

$$\ddot{\bar{u}} = -\omega^2 \bar{q} \exp(j\omega t) \quad (34)$$

By substitution,

$$-\omega^2 \hat{m}_{ww} \bar{q} \exp(j\omega t) + \hat{K}_{ww} \bar{q} \exp(j\omega t) = \bar{0} \quad (35)$$

$$\left\{ -\omega^2 \hat{m}_{ww} \bar{q} + \hat{K}_{ww} \bar{q} \right\} \exp(j\omega t) = \bar{0} \quad (36)$$

$$\left\{ -\omega_n^2 \hat{m}_{ww} \bar{q} + \hat{K}_{ww} \bar{q} \right\} \exp(j\omega_n t) = \bar{0} \quad (37)$$

$$\left\{ -\omega^2 \hat{m}_{ww} + \hat{K}_{ww} \right\} \bar{q} = \bar{0} \quad (38)$$

$$\left\{ \hat{K}_{ww} - \omega^2 \hat{m}_{ww} \right\} \bar{q} = \bar{0} \quad (39)$$

Equation (39) is an example of a generalized eigenvalue problem. The eigenvalues can be found by setting the determinant equal to zero.

$$\det \left\{ \hat{K}_{ww} - \omega^2 \hat{m}_{ww} \right\} = 0 \quad (40)$$

The eigenvectors are found via the following equations.

$$\left\{ \hat{\mathbf{K}}_{\text{ww}} - \omega_i^2 \hat{\mathbf{m}}_{\text{ww}} \right\} \bar{\mathbf{q}}_i = \bar{\mathbf{0}} \quad (41)$$

An eigenvector matrix  $\mathbf{Q}$  can be formed. The eigenvectors are inserted in column format.

$$\mathbf{Q} = [\bar{\mathbf{q}}_1 \mid \bar{\mathbf{q}}_2 \mid \dots \mid \bar{\mathbf{q}}_n] \quad (42)$$

where  $n$  is the number of degrees-of-freedom

The eigenvectors represent orthogonal mode shapes. Assume that the eigenvectors are mass-normalized such that

$$\mathbf{Q}^T \mathbf{M} \mathbf{Q} = \mathbf{I} \quad (43)$$

and

$$\mathbf{Q}^T \mathbf{K} \mathbf{Q} = \mathbf{\Omega} \quad (44)$$

where

superscript  $T$  represents transpose

$\mathbf{I}$  is the identity matrix

$\mathbf{\Omega}$  is a diagonal matrix of eigenvalues

Now define a modal coordinate  $\eta(t)$  such that

$$\bar{\mathbf{u}} = \mathbf{Q} \bar{\boldsymbol{\eta}} \quad (45)$$

Let  $q_{ij}$  represent the elements of  $\mathbf{Q}$ .

The displacement, velocity and acceleration terms are

$$u_i = \sum_{j=1}^n q_{ij} \eta_j \quad (46)$$

$$\dot{u}_i = \sum_{j=1}^n q_{ij} \dot{\eta}_j \quad (47)$$

$$\ddot{u}_i = \sum_{j=1}^n q_{ij} \ddot{\eta}_j \quad (48)$$

By substitution

$$\hat{m}_{ww} Q \ddot{\eta} + \hat{K}_{ww} Q \eta = -\hat{m}_{wd} \ddot{u}_d \quad (49)$$

Premultiply by the transpose of the normalized eigenvector matrix.

$$Q^T \hat{m}_{ww} Q \ddot{\eta} + Q^T \hat{K}_{ww} Q \eta = -\hat{Q}^T \hat{m}_{wd} \ddot{u}_d \quad (50)$$

The orthogonality relationships yield

$$I \ddot{\eta} + \Omega \eta = -\hat{Q}^T \hat{m}_{wd} \ddot{u}_d \quad (51)$$

Note that the two equations are decoupled in terms of the modal coordinate.

Now assume modal damping by adding an uncoupled damping matrix.

$$I \ddot{\eta} + D \dot{\eta} + \Omega \eta = -\hat{Q}^T \hat{m}_{wd} \ddot{u}_d \quad (52)$$

$$D_{ij} = \begin{cases} 2\xi_i \omega_i^2, & \text{for } i = j \\ 0, & \text{for } i \neq j \end{cases} \quad (53)$$



Now assume a harmonic base input. Assume that it is uniform if it is applied at multiple locations.

$$\ddot{y} = A \exp(j\omega t) \quad (54)$$

Assume a common harmonic modal displacement.

$$\eta_i = \psi_i \exp(j\omega t) \quad (55)$$

$$\dot{\eta}_i = j\omega_i \psi_i \exp(j\omega t) \quad (56)$$

$$\ddot{\eta}_i = -\omega_i^2 \psi_i \exp(j\omega t) \quad (57)$$

Let  $C$  be a vectors of ones. The number of rows in  $C$  is equal to the number of drive points.

By substitution,

$$\left\{ -\omega^2 + j2\xi_i \omega_i \omega + \omega_i^2 \right\} \psi_i \exp(j\omega t) = - \left\{ \left[ Q^T \hat{m}_{wd} \right]_{row i} C \right\} A \exp(j\omega t) \quad (58)$$

$$\left\{ \left[ \omega_i^2 - \omega^2 \right] + j2\xi_i \omega_i \omega \right\} \psi_i \exp(j\omega t) = - \left\{ \left[ Q^T \hat{m}_{wd} \right]_{row i} C \right\} A \exp(j\omega t) \quad (59)$$

The modal displacement is

$$\eta_i = \psi_i \exp(j\omega t) = \frac{- \left\{ \left[ Q^T \hat{m}_{wd} \right]_{row i} C \right\}}{\left\{ \left[ \omega_i^2 - \omega^2 \right] + j2\xi_i \omega_i \omega \right\}} A \exp(j\omega t) \quad (60)$$

The modal velocity is

$$\dot{\eta}_i = \frac{-j\omega \left\{ \left[ Q^T \hat{m}_{wd} \right]_{row i} C \right\}}{\left\{ \left[ \omega_i^2 - \omega^2 \right] + j 2 \xi_i \omega_i \omega \right\}} A \exp(j\omega t) \quad (61)$$

The modal acceleration is

$$\ddot{\eta}_i = \frac{\omega^2 \left\{ \left[ Q^T \hat{m}_{wd} \right]_{row i} C \right\}}{\left\{ \left[ \omega_i^2 - \omega^2 \right] + j 2 \xi_i \omega_i \omega \right\}} A \exp(j\omega t) \quad (62)$$

Recall

$$\ddot{u}_i = \sum_{p=1}^n q_{ip} \ddot{\eta}_p \quad (63)$$

$$\ddot{u}_i = \sum_{p=1}^n \left\{ q_{ip} \frac{\omega^2 \left\{ \left[ Q^T \hat{m}_{wd} \right]_{row i} C \right\}}{\left\{ \left[ \omega_p^2 - \omega^2 \right] + j 2 \xi_p \omega_p \omega \right\}} A \exp(j\omega t) \right\} \quad (64)$$

$$\ddot{u}_i = A \exp(j\omega t) \sum_{p=1}^n \left\{ q_{ip} \frac{\omega^2 \left\{ \left[ Q^T \hat{m}_{wd} \right]_{row i} C \right\}}{\left\{ \left[ \omega_p^2 - \omega^2 \right] + j 2 \xi_p \omega_p \omega \right\}} \right\} \quad (65)$$

The Fourier transform equation is

$$\hat{U}_i(f) = \int_{-\infty}^{\infty} \ddot{u}_i(t) \exp[-j\omega t] dt \quad (66)$$

$$\ddot{u}_i = A \exp(j\omega t) \sum_{p=1}^n \left\{ q_{ip} \frac{\omega^2 \left\{ \left[ Q^T \hat{m}_{wd} \right]_{row i} C \right\}}{\left[ \omega_p^2 - \omega^2 \right] + j 2 \xi_p \omega_p \omega} \right\} \quad (67)$$

$$\hat{U}_i(f)/A = \sum_{p=1}^n \left\{ q_{ip} \frac{\omega^2 \left\{ \left[ Q^T \hat{m}_{wd} \right]_{row i} C \right\}}{\left[ \omega_p^2 - \omega^2 \right] + j 2 \xi_p \omega_p \omega} \right\} \quad (68)$$

Recall

$$\begin{bmatrix} u_d \\ u_f \end{bmatrix} = \Pi \begin{bmatrix} u_d \\ u_w \end{bmatrix} \quad (69)$$

The equivalent format for the frequency domain is

$$\begin{bmatrix} \hat{U}_d \\ \hat{U}_f \end{bmatrix} = \Pi \begin{bmatrix} \hat{U}_d \\ \hat{U}_w \end{bmatrix} \quad (70)$$

The final step is to rearrange the degrees-of-freedom in the proper order.

## References

1. T. Irvine, Modal Transient Analysis of a Multi-degree-of-freedom System with Enforced Motion, Revision E, Vibrationdata, 2012.
2. T. Irvine, Transverse Vibration of a Beam via the Finite Element Method, Revision F, Vibrationdata, 2010.

## APPENDIX A

### Example

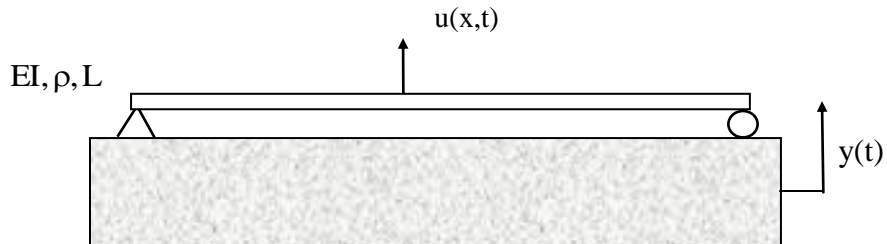


Figure A-1.

Consider a beam subjected to uniform base excitation at each end.

The beam has the following properties:

Cross-Section	Circular
Boundary Conditions	Simply-Supported at Each End
Material	Aluminum

Diameter	D	=	0.5 inch
Cross-Section Area	A	=	0.1963 in <sup>2</sup>
Length	L	=	24 inch
Area Moment of Inertia	I	=	0.003068 in <sup>4</sup>
Elastic Modulus	E	=	1.0e+07 lbf/in <sup>2</sup>
Stiffness	EI	=	30680 lbf in <sup>2</sup>
Mass per Volume	$\rho_v$	=	0.1 lbm / in <sup>3</sup> ( 0.000259 lbf sec <sup>2</sup> /in <sup>4</sup> )
Mass per Length	$\rho$	=	0.01963 lbm/in (5.08e-05 lbf sec <sup>2</sup> /in <sup>2</sup> )
Mass	$\rho L$	=	0.471 lbm (1.22E-03 lbf sec <sup>2</sup> /in)
Viscous Damping Ratio	$\xi$	=	0.05

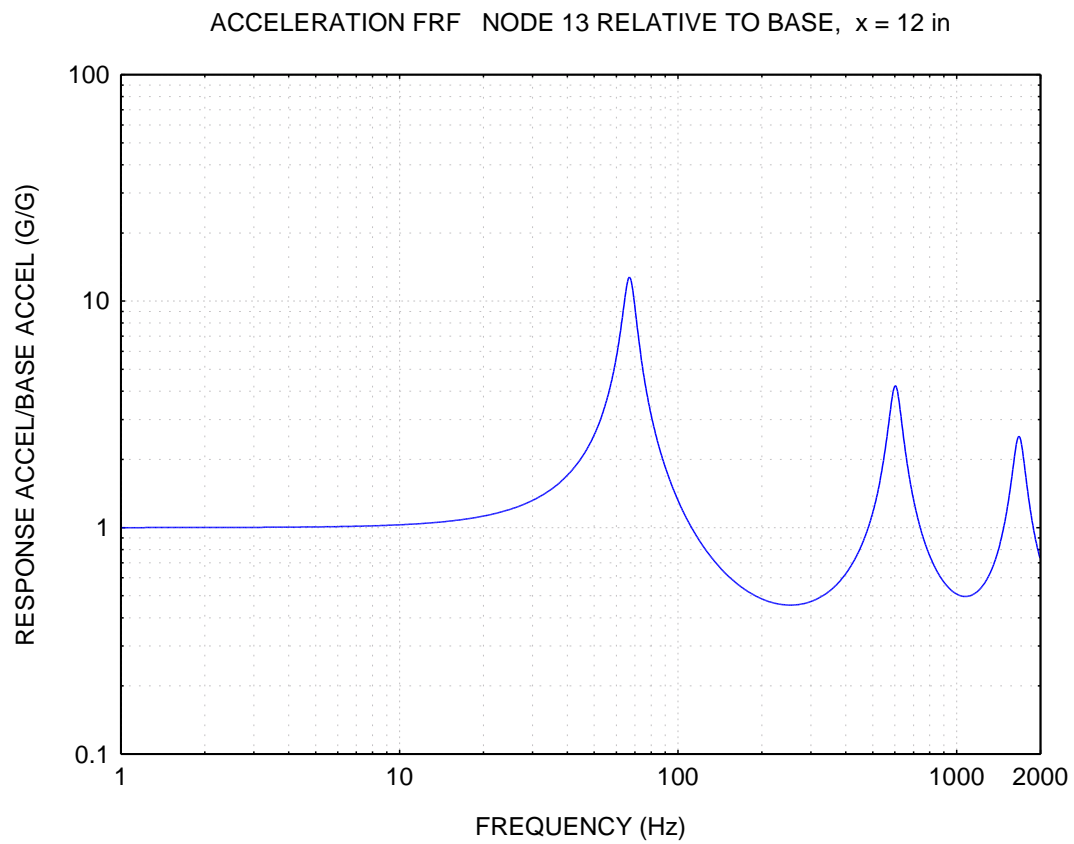


Figure A-2.

Model the beam with 24 elements using the method in Reference 2. Calculate the frequency response function at the center.