

MULTI-DEGREE-OF-FREEDOM SYSTEM  
SUBJECTED TO AN ARBITRARY FORCE  
Revision A

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Two-degree-of-freedom System

A two-degree-of-freedom system is used as an example, as shown in Figure 1.

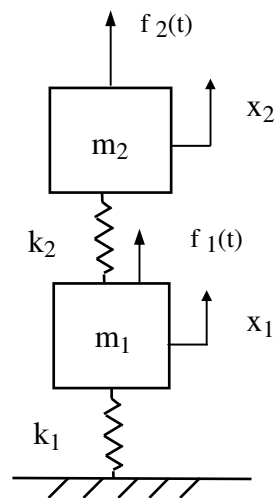


Figure 1.

A free-body diagram of mass 1 is given in Figure 2. A free-body diagram of mass 2 is given in Figure 3.

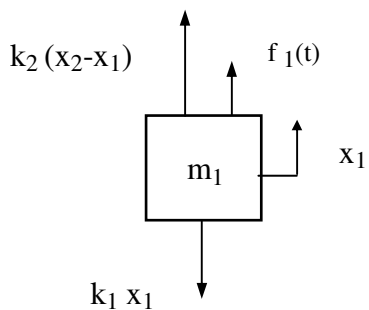


Figure 2.

Determine the equation of motion for mass 1.

$$\Sigma F = m_1 \ddot{x}_1 \tag{1}$$

$$m_1 \ddot{x}_1 = f_1(t) + k_2(x_2 - x_1) - k_1 x_1 \tag{2}$$

$$m_1 \ddot{x}_1 + k_1 x_1 - k_2(x_2 - x_1) = f_1(t) \tag{3}$$

$$m_1 \ddot{x}_1 + k_1 x_1 + k_2(x_1 - x_2) = f_1(t) \tag{4}$$

$$m_1 \ddot{x}_1 + (k_1 + k_2)x_1 - k_2 x_2 = f_1(t) \tag{5}$$

$m_2$

$k_2(x_2 - x_1)$

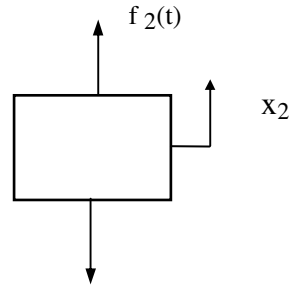


Figure 3.

Derive the equation of motion for mass 2.

$$\Sigma F = m_2 \ddot{x}_2 \quad (6)$$

$$m_2 \ddot{x}_2 = f_2(t) - k_2(x_2 - x_1) \quad (7)$$

$$m_2 \ddot{x}_2 + k_2(x_2 - x_1) = f_2(t) \quad (8)$$

$$m_2 \ddot{x}_2 + k_2 x_2 - k_2 x_1 = f_2(t) \quad (9)$$

Assemble the equations in matrix form.

$$\begin{bmatrix} m_1 & 0 \\ 0 & m_2 \end{bmatrix} \begin{bmatrix} \ddot{x}_1 \\ \ddot{x}_2 \end{bmatrix} + \begin{bmatrix} k_1 + k_2 & -k_2 \\ -k_2 & k_2 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} = \begin{bmatrix} f_1(t) \\ f_2(t) \end{bmatrix} \quad (10)$$

### Decoupling

Equation (10) is coupled via the stiffness matrix. An intermediate goal is to decouple the equation.

Simplify,

$$\mathbf{M} \ddot{\bar{x}} + \mathbf{K} \bar{x} = \bar{F} \quad (11)$$

where

$$\mathbf{M} = \begin{bmatrix} m_1 & 0 \\ 0 & m_2 \end{bmatrix} \quad (12)$$

$$\mathbf{K} = \begin{bmatrix} k_1 + k_2 & -k_2 \\ -k_2 & k_2 \end{bmatrix} \quad (13)$$

$$\bar{\mathbf{x}} = \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} \quad (14)$$

$$\bar{\mathbf{F}} = \begin{bmatrix} f_1(t) \\ f_2(t) \end{bmatrix} \quad (15)$$

Consider the homogeneous form of equation (11).

$$\mathbf{M} \ddot{\bar{\mathbf{x}}} + \mathbf{K} \bar{\mathbf{x}} = \bar{\mathbf{0}} \quad (16)$$

Seek a solution of the form

$$\bar{\mathbf{x}} = \bar{\mathbf{q}} \exp(j\omega t) \quad (17)$$

The  $\mathbf{q}$  vector is the generalized coordinate vector.

Note that

$$\dot{\bar{\mathbf{x}}} = j\omega \bar{\mathbf{q}} \exp(j\omega t) \quad (18)$$

$$\ddot{\bar{\mathbf{x}}} = -\omega^2 \bar{\mathbf{q}} \exp(j\omega t) \quad (19)$$

Substitute equations (17) through (19) into equation (16).

$$-\omega^2 \mathbf{M} \bar{\mathbf{q}} \exp(j\omega t) + \mathbf{K} \bar{\mathbf{q}} \exp(j\omega t) = \bar{\mathbf{0}} \quad (20)$$

$$\{-\omega^2 \mathbf{M} \bar{\mathbf{q}} + \mathbf{K} \bar{\mathbf{q}}\} \exp(j\omega t) = \bar{\mathbf{0}} \quad (21)$$

$$-\omega_n^2 \mathbf{M} \bar{\mathbf{q}} + \mathbf{K} \bar{\mathbf{q}} = \bar{\mathbf{0}} \quad (22)$$

$$\{-\omega^2 \mathbf{M} + \mathbf{K}\} \bar{\mathbf{q}} = \bar{\mathbf{0}} \quad (23)$$

$$\{K - \omega^2 M\} \bar{q} = \bar{0} \quad (24)$$

Equation (24) is an example of a generalized eigenvalue problem. The eigenvalues can be found by setting the determinant equal to zero.

$$\det \{K - \omega^2 M\} = 0 \quad (25)$$

$$\det \left\{ \begin{bmatrix} k_1 + k_2 & -k_2 \\ -k_2 & k_2 \end{bmatrix} - \omega^2 \begin{bmatrix} m_1 & 0 \\ 0 & m_2 \end{bmatrix} \right\} = 0 \quad (26)$$

$$\det \begin{bmatrix} (k_1 + k_2) - \omega^2 m_1 & -k_2 \\ -k_2 & k_2 - \omega^2 m_2 \end{bmatrix} = 0 \quad (27)$$

$$\left[ (k_1 + k_2) - \omega^2 m_1 \right] \left[ k_2 - \omega^2 m_2 \right] - k_2^2 = 0 \quad (28)$$

$$\omega^4 m_1 m_2 - \omega^2 [m_1 k_2 + m_2 (k_1 + k_2)] - k_2^2 = 0 \quad (29)$$

The eigenvalues are the roots of the polynomial.

$$\omega_1^2 = \frac{-b - \sqrt{b^2 - 4ac}}{2a} \quad (30)$$

$$\omega_2^2 = \frac{-b + \sqrt{b^2 - 4ac}}{2a} \quad (31)$$

where

$$a = m_1 m_2 \quad (32)$$

$$b = -[m_1 k_2 + m_2 (k_1 + k_2)] \quad (33)$$

$$c = -k_2^2 \quad (34)$$

The eigenvectors are found via the following equations.

$$\left\{ \mathbf{K} - \omega_1^2 \mathbf{M} \right\} \bar{\mathbf{q}}_1 = \bar{\mathbf{0}} \quad (35)$$

$$\left\{ \mathbf{K} - \omega_2^2 \mathbf{M} \right\} \bar{\mathbf{q}}_2 = \bar{\mathbf{0}} \quad (36)$$

where

$$\bar{\mathbf{q}}_1 = \begin{bmatrix} v_1 \\ v_2 \end{bmatrix} \quad (37)$$

$$\bar{\mathbf{q}}_2 = \begin{bmatrix} w_1 \\ w_2 \end{bmatrix} \quad (38)$$

An eigenvector matrix  $\mathbf{Q}$  can be formed. The eigenvectors are inserted in column format.

$$\mathbf{Q} = [ \bar{\mathbf{q}}_1 \quad | \quad \bar{\mathbf{q}}_2 ] \quad (39)$$

$$\mathbf{Q} = \begin{bmatrix} v_1 & w_1 \\ v_2 & w_2 \end{bmatrix} \quad (40)$$

The eigenvectors represent orthogonal mode shapes.

Each eigenvector can be multiplied by an arbitrary scale factor. A mass-normalized eigenvector matrix  $\hat{\mathbf{Q}}$  can be obtained such that the following orthogonality relations are obtained.

$$\hat{\mathbf{Q}}^T \mathbf{M} \hat{\mathbf{Q}} = \mathbf{I} \quad (41)$$

and

$$\hat{\mathbf{Q}}^T \mathbf{K} \hat{\mathbf{Q}} = \mathbf{\Omega} \quad (42)$$

where

- T    Superscript, represents transpose
- I    is the identity matrix
- $\mathbf{\Omega}$     is a diagonal matrix of eigenvalues

Note that

$$\hat{Q} = \begin{bmatrix} \hat{v}_1 & \hat{w}_1 \\ \hat{v}_2 & \hat{w}_2 \end{bmatrix} \quad (43a)$$

$$\hat{Q}^T = \begin{bmatrix} \hat{v}_1 & \hat{v}_2 \\ \hat{w}_1 & \hat{w}_2 \end{bmatrix} \quad (43b)$$

Rigorous proof of the orthogonality relationships is beyond the scope of this tutorial. Further discussion is given in References 1 and 2.

Nevertheless, the orthogonality relationships are demonstrated by an example in this tutorial.

Now define a generalized coordinate  $\eta(t)$  such that

$$\bar{x} = \hat{Q} \bar{\eta} \quad (44)$$

Substitute equation (43) into the equation of motion, equation (11).

$$M \hat{Q} \ddot{\bar{\eta}} + K \hat{Q} \bar{\eta} = \bar{F} \quad (45)$$

Premultiply by the transpose of the normalized eigenvector matrix.

$$\hat{Q}^T M \hat{Q} \ddot{\bar{\eta}} + \hat{Q}^T K \hat{Q} \bar{\eta} = \hat{Q}^T \bar{F} \quad (46)$$

The orthogonality relationships yield

$$I \ddot{\bar{\eta}} + \Omega \bar{\eta} = \hat{Q}^T \bar{F} \quad (47)$$

For the sample problem, equation (46) becomes

$$\begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix} \begin{bmatrix} \ddot{\eta}_1 \\ \ddot{\eta}_2 \end{bmatrix} + \begin{bmatrix} \omega_1^2 & 0 \\ 0 & \omega_2^2 \end{bmatrix} \begin{bmatrix} \eta_1 \\ \eta_2 \end{bmatrix} = \begin{bmatrix} \hat{v}_1 & \hat{v}_2 \\ \hat{w}_1 & \hat{w}_2 \end{bmatrix} \begin{bmatrix} f_1(t) \\ f_2(t) \end{bmatrix} \quad (48)$$

Note that the two equations are decoupled in terms of the generalized coordinate.

Add modal damping.

$$\begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix} \begin{bmatrix} \dot{\eta}_1 \\ \dot{\eta}_2 \end{bmatrix} + \begin{bmatrix} 2\xi_1\omega_1 & 0 \\ 0 & 2\xi_2\omega_2 \end{bmatrix} \begin{bmatrix} \dot{\eta}_1 \\ \dot{\eta}_2 \end{bmatrix} + \begin{bmatrix} \omega_1^2 & 0 \\ 0 & \omega_2^2 \end{bmatrix} \begin{bmatrix} \eta_1 \\ \eta_2 \end{bmatrix} = \begin{bmatrix} \hat{v}_1 & \hat{v}_2 \\ \hat{w}_1 & \hat{w}_2 \end{bmatrix} \begin{bmatrix} f_1(t) \\ f_2(t) \end{bmatrix} \quad (49)$$

Equation (49) yields two equations

$$\ddot{\eta}_1 + 2\xi_1\omega_1\dot{\eta}_1 + \omega_1^2 \eta_1 = \hat{v}_1 f_1(t) + \hat{v}_2 f_2(t) \quad (50)$$

$$\ddot{\eta}_2 + 2\xi_2\omega_2\dot{\eta}_2 + \omega_2^2 \eta_2 = \hat{w}_1 f_1(t) + \hat{w}_2 f_2(t) \quad (51)$$

The equations can be solved in terms of Laplace transforms, or some other differential equation solution method.

Now consider the initial conditions. Recall

$$\bar{x} = \hat{Q} \bar{\eta} \quad (52)$$

Thus,

$$\bar{x}(0) = \hat{Q} \bar{\eta}(0) \quad (53)$$

Premultiply by  $\hat{Q}^T M$ .

$$\hat{Q}^T M \bar{x}(0) = \hat{Q}^T M \hat{Q} \eta(0) \quad (54)$$

Recall

$$\hat{Q}^T M \hat{Q} = I \quad (55)$$

$$\hat{Q}^T M \bar{x}(0) = I \eta(0) \quad (56)$$

$$\hat{Q}^T M \bar{x}(0) = \eta(0) \quad (57)$$

Finally, the transformed initial displacement is

$$\eta(0) = \hat{Q}^T M \bar{x}(0) \quad (58)$$



Similarly, the transformed initial velocity is

$$\dot{\eta}(0) = \hat{Q}^T M \bar{x}(0) \quad (59)$$

A basis for a solution is thus derived.

### Arbitrary Force

Now consider the case of arbitrary forcing functions.

$$\ddot{\eta}_1 + 2\xi_1\omega_1\dot{\eta}_1 + \omega_1^2 \eta_1 = \hat{v}_1 f_1(t) + \hat{v}_2 f_2(t) \quad (60)$$

$$\ddot{\eta}_2 + 2\xi_2\omega_2\dot{\eta}_2 + \omega_2^2 \eta_2 = \hat{w}_1 f_1(t) + \hat{w}_2 f_2(t) \quad (61)$$

Apply the convolution integral per Reference 1.

$$\begin{aligned} \eta_1(t) = & -\frac{\hat{v}_1}{\omega_{d,1}} \int_0^t f_1(\tau) \left[ \exp\{-\xi_1\omega_{n,1}(t-\tau)\} \right] \left[ \sin\omega_{d,1}(t-\tau) \right] d\tau \\ & -\frac{\hat{v}_2}{\omega_{d,1}} \int_0^t f_2(\tau) \left[ \exp\{-\xi_1\omega_{n,1}(t-\tau)\} \right] \left[ \sin\omega_{d,1}(t-\tau) \right] d\tau \end{aligned} \quad (62)$$

$$\begin{aligned} \eta_2(t) = & -\frac{\hat{w}_1}{\omega_{d,2}} \int_0^t f_1(\tau) \left[ \exp\{-\xi_2\omega_{n,2}(t-\tau)\} \right] \left[ \sin\omega_{d,2}(t-\tau) \right] d\tau \\ & -\frac{\hat{w}_2}{\omega_{d,2}} \int_0^t f_2(\tau) \left[ \exp\{-\xi_2\omega_{n,2}(t-\tau)\} \right] \left[ \sin\omega_{d,2}(t-\tau) \right] d\tau \end{aligned} \quad (63)$$

$$\begin{aligned}
\eta_{1,i} = & + \{2 \exp[-\xi_1 \omega_{n,1} \Delta t] \cos[\omega_{d,1} \Delta t] \eta_{i-1}\} \\
& - \{\exp[-2\xi_1 \omega_{n,1} \Delta t] \eta_{i-2}\} \\
& + \{\hat{v}_1 f_{1,i-1} + \hat{v}_2 f_{2,i-1}\} \{\Delta t / \omega_{d,1}\} \exp[-\xi_1 \omega_{n,1} \Delta t] \{\sin[\omega_{d,1} \Delta t]\}
\end{aligned} \tag{64}$$

$$\begin{aligned}
\eta_{2,i} = & + \{2 \exp[-\xi_2 \omega_{n,2} \Delta t] \cos[\omega_{d,2} \Delta t] \eta_{i-1}\} \\
& - \{\exp[-2\xi_2 \omega_{n,2} \Delta t] \eta_{i-2}\} \\
& + \{\hat{v}_1 f_{1,i-1} + \hat{v}_2 f_{2,i-1}\} \{\Delta t / \omega_{d,2}\} \exp[-\xi_2 \omega_{n,2} \Delta t] \{\sin[\omega_{d,2} \Delta t]\}
\end{aligned} \tag{65}$$

The modal velocity is

$$\begin{aligned}
\dot{\eta}_{1,i} = & + \{2 \exp[-\xi_1 \omega_{n,1} \Delta t] \cos[\omega_{d,1} \Delta t] \dot{\eta}_{i-1}\} \\
& - \{\exp[-2\xi_1 \omega_{n,1} \Delta t] \dot{\eta}_{i-2}\} \\
& + \Delta t \{\hat{v}_1 f_{1,i} + \hat{v}_2 f_{2,i}\} \\
& + \{\hat{v}_1 f_{1,i-1} + \hat{v}_2 f_{2,i-1}\} \{\Delta t\} \exp[-\xi_1 \omega_{n,1} \Delta t] \left\{ \left[ \frac{-\xi_1 \omega_{n,1}}{\omega_{d,1}} \right] \sin[\omega_{d,1} \Delta t] - \cos[\omega_{d,1} \Delta t] \right\}
\end{aligned} \tag{66}$$

$$\begin{aligned}
\dot{\eta}_{2,i} = & + \{2 \exp[-\xi_2 \omega_{n,2} \Delta t] \cos[\omega_{d,2} \Delta t] \dot{\eta}_{i-1}\} \\
& - \{\exp[-2\xi_2 \omega_{n,2} \Delta t] \dot{\eta}_{i-2}\} \\
& + \Delta t \{\hat{w}_1 f_{1,i} + \hat{w}_2 f_{2,i}\} \\
& + \{\hat{w}_1 f_{1,i-1} + \hat{w}_2 f_{2,i-1}\} \{\Delta t\} \exp[-\xi_1 \omega_{n,1} \Delta t] \left\{ \left[ \frac{-\xi_2 \omega_{n,2}}{\omega_{d,2}} \right] \sin[\omega_{d,2} \Delta t] - \cos[\omega_{d,2} \Delta t] \right\}
\end{aligned} \tag{67}$$

The acceleration values can then be found via

$$\ddot{\eta}_1 = -2\xi_1 \omega_1 \dot{\eta}_1 - \omega_1^2 \eta_1 + \hat{v}_1 f_1(t) + \hat{v}_2 f_2(t) \tag{68}$$

$$\ddot{\eta}_2 = -2\xi_2 \omega_2 \dot{\eta}_2 - \omega_2^2 \eta_2 + \hat{w}_1 f_1(t) + \hat{w}_2 f_2(t) \tag{69}$$

The physical displacements are then found as

$$\bar{x} = \hat{Q} \bar{\eta} \tag{70}$$

$$\begin{bmatrix} x_1(t) \\ x_2(t) \end{bmatrix} = \begin{bmatrix} \hat{v}_1 & \hat{w}_1 \\ \hat{v}_2 & \hat{w}_2 \end{bmatrix} \begin{bmatrix} \eta_1(t) \\ \eta_2(t) \end{bmatrix} \tag{71}$$

The velocity is

$$\dot{\bar{x}} = \hat{Q} \dot{\bar{\eta}} \tag{72}$$

The acceleration is

$$\ddot{\bar{x}} = \hat{Q} \ddot{\bar{\eta}} \tag{73}$$

## Reference

1. T. Irvine, An Introduction to the Shock Response Spectrum Rev P, Vibrationdata, 2002.

## APPENDIX A

### Example

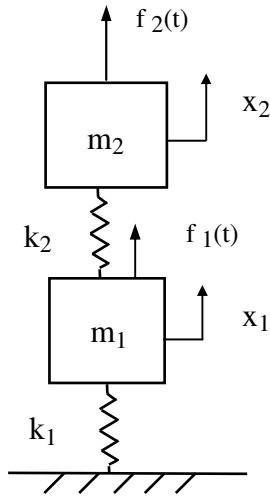


Figure A-1.

Let

1. Each mass = 1 lbm
2. Each spring = 2500 lbf/in
3. Each modal damping = 0.05
4.  $f_1 = 0$
5.  $f_2 = 1$  lbf, 0.010 sec, half-sine pulse

The response is calculated using the formulas derived in the main text as implemented via Matlab script: `mdof_arb_force.m`

The response time histories are shown in the following figures.

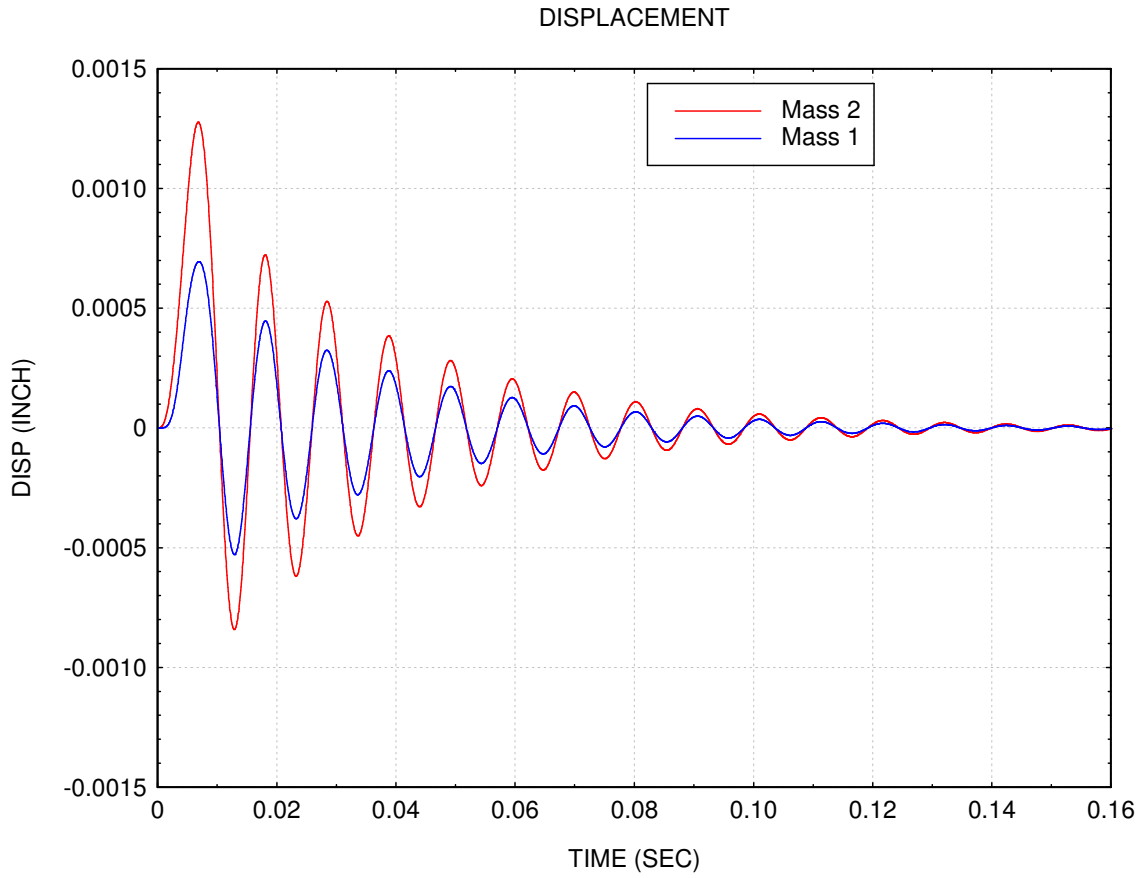


Figure A-2.

Natural Frequencies =

96.63 Hz  
253 Hz

Modes Shapes (column format) =

-10.33	-16.71
-16.71	10.33

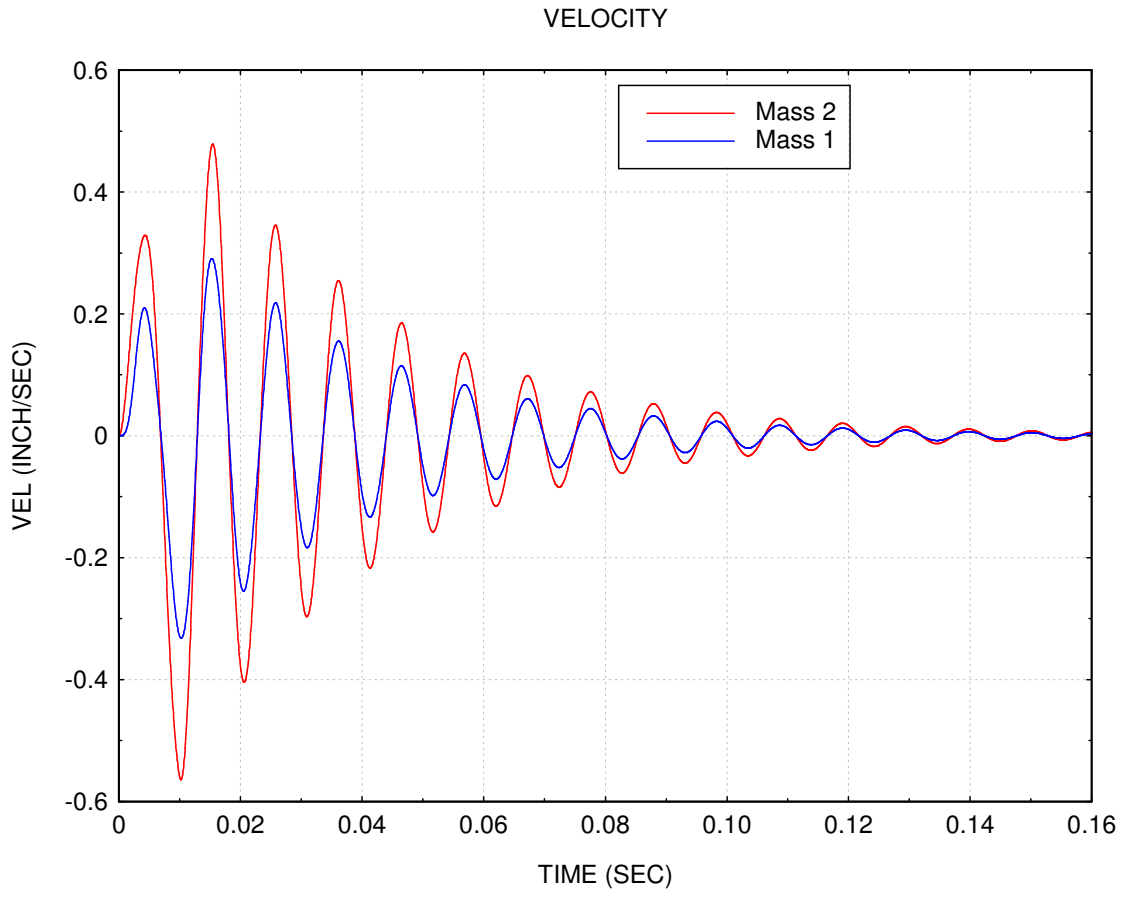


Figure A-3.

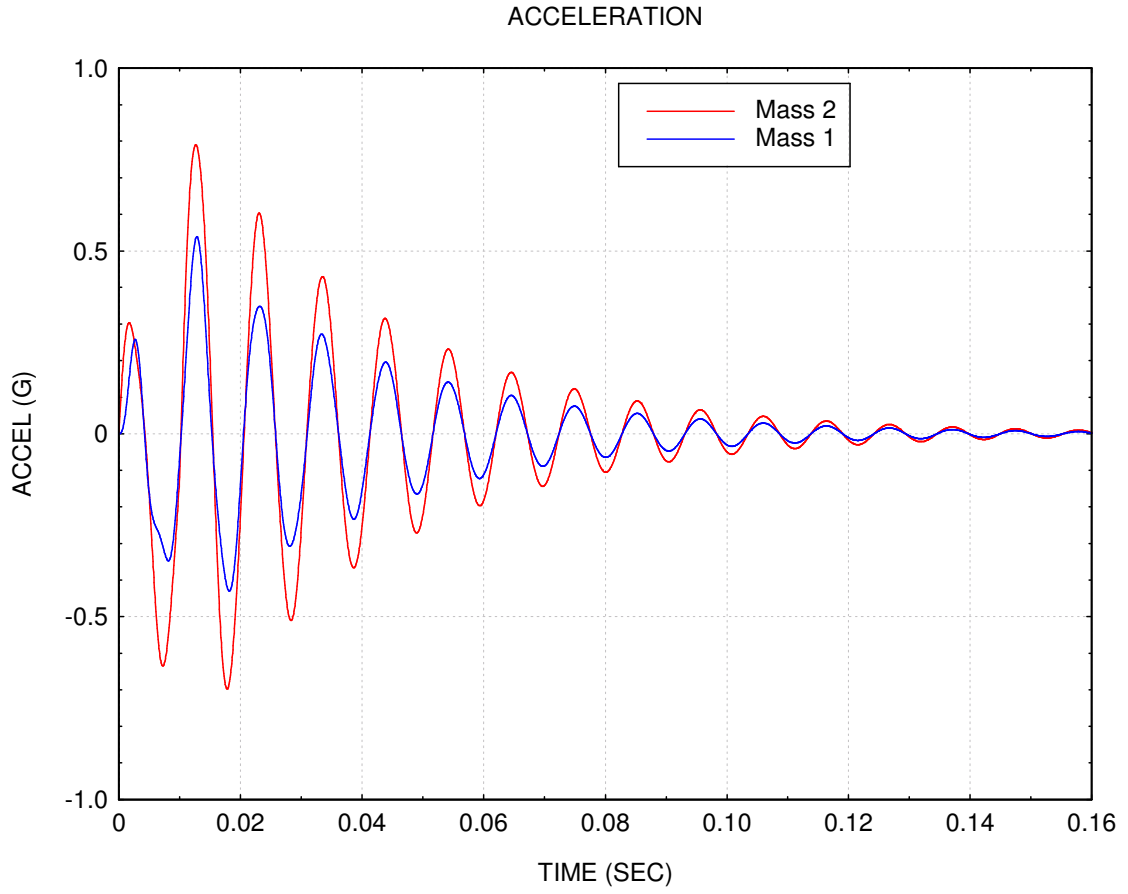


Figure A-4.